

# Radiation Hardness Assurance (RHA): Challenges and New Considerations

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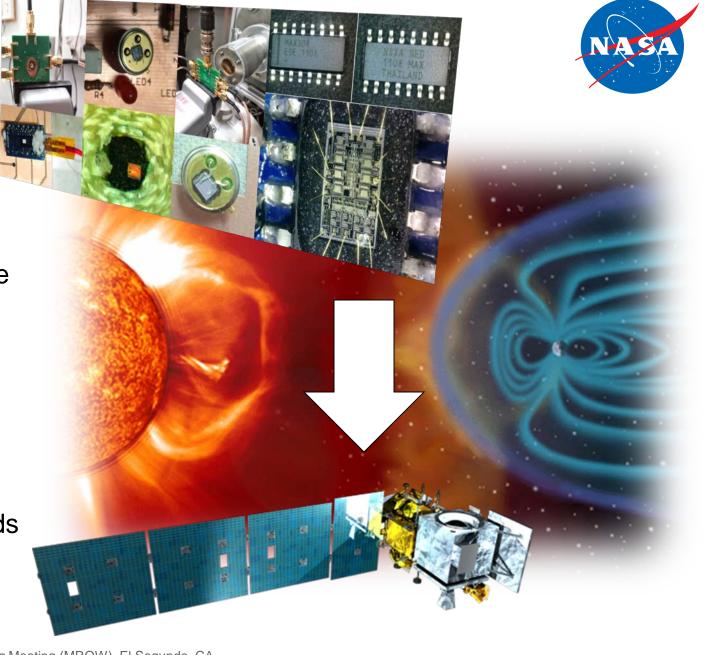
# Acronyms



| COTS | Commercial Off The Shelf            |  |  |  |
|------|-------------------------------------|--|--|--|
| DD   | Displacement Damage                 |  |  |  |
| GEO  | Geostationary Earth Orbit           |  |  |  |
| GSFC | Goddard Space Flight Center         |  |  |  |
| LEO  | Low Earth Orbit                     |  |  |  |
| LET  | Linear Energy Transfer              |  |  |  |
| MBU  | Multi-Bit Upset                     |  |  |  |
| MCU  | Multi-Cell Upset                    |  |  |  |
| NEPP | NASA Electronic Parts and Packaging |  |  |  |
| RDM  | Radiation Design Margin             |  |  |  |
| RHA  | Radiation Hardness Assurance        |  |  |  |
| SEB  | Single Event Burnout                |  |  |  |
| SEDR | Single Event Dielectric Rupture     |  |  |  |
| SEE  | Single Event Effects                |  |  |  |
| SEFI | Single Event Functional Interrupt   |  |  |  |
| SEGR | Single Event Gate Rupture           |  |  |  |
| SEL  | Single Event Latchup                |  |  |  |
| SOA  | Safe Operating Area                 |  |  |  |
| TID  | Total Ionizing Dose                 |  |  |  |
|      |                                     |  |  |  |

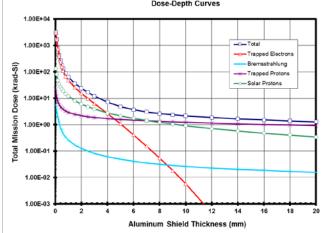
# RHA Challenges

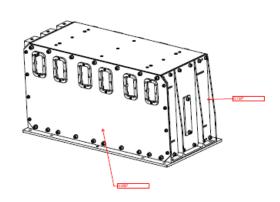
- New Technologies
  - Radiation Testing / Data collection
  - Modeling the Physics of Failure
- Test Facility Access
- Increased COTS parts/subsystem usage
- Translation of system requirements into radiation pass/fail criteria
- Determining appropriate mitigation level (operational, system, circuit/software, device, material, etc.)
- Wide range of mission profiles and needs
- Always in a <u>dynamic</u> environment

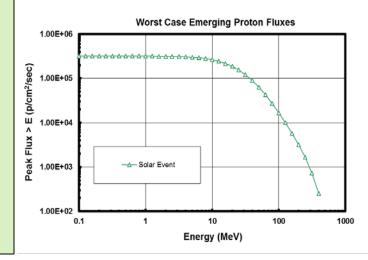


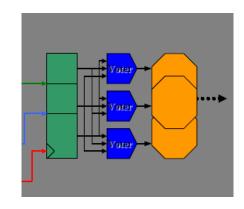
# RHA Flow Doesn't Change With Risk or Mission

- **Define the Environment** 
  - External to the spacecraft
- **Evaluate the Environment** 
  - Internal to the spacecraft
- **Define the Requirements** 
  - Define criticality factors
- **Evaluate Design/Components** 
  - Existing data/Testing
  - Performance characteristics
- "Engineer" with Designers
  - Parts replacement/Mitigation schemes
- **Iterate Process** 
  - Review parts list based on updated knowledge







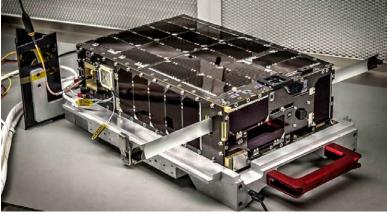


K.A. LaBel, A.H. Johnston, J.L. Barth, R.A. Reed, C.E. Barnes, "Emerging Radiation Hardness Assurance (RHA) issues: A NASA approach for space flight programs," IEEE Trans. Nucl. Sci., pp. 2727-2736, Dec. 1998.

# Risk Acceptance Will Change

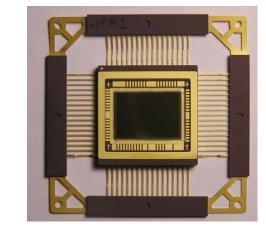


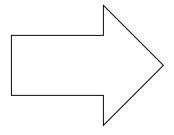
- Mission Profiles Are Expanding
  - Based on mission life, objective, and cost
  - Oversight gives way to insight for lower class
  - o Ground systems, do no harm, hosted payloads
  - Similarity and heritage data requirement widening
  - o In some cases unbounded radiation risks are likely



Credits: NASA's Goddard Space Flight Center/Bill Hrybyk

Part Classifications Growing







# Summary of Environmental Hazards



|  | Plasma<br>(charging)  | Trapped<br>Protons  | Trapped<br>Electrons  | Solar Particles  | Cosmic Rays | Human<br>Presence | Long Lifetime<br>(>10 years) | Nuclear<br>Exposure | Repeated<br>Launch | Extreme<br>Temperature | Planetary<br>Contaminates<br>(Dust, etc) |
|--|---|---|---|------------------|-------------|-------------------|------------------------------|---------------------|--------------------|------------------------|--|
| GEO                                      | Yes   | No  | Severe  | Yes              | Yes         | No                | Yes                          | No                  | No                 | No                     | No                                       |
| LEO (low-incl)                           | No  | Yes   | Moderate  | No               | No          | No                | Not<br>usual                 | No                  | No                 | No                     | No                                       |
| LEO Polar                                | No  | Yes   | Moderate  | Yes              | Yes         | No                | Not<br>usual                 | No                  | No                 | No                     | No                                       |
| ISS                                      | No  | Yes   | Moderate  | Yes -<br>partial | Minimal     | Yes               | Yes                          | No                  | Yes                | No                     | No                                       |
| Interplanetary                           | During<br>phasing<br>orbits;<br>Possible<br>Other<br>Planet | During<br>phasing<br>orbits;<br>Possible<br>Other<br>Planet | During<br>phasing<br>orbits;<br>Possible<br>Other<br>Planet | Yes              | Yes         | No                | Yes                          | Maybe               | No                 | Yes                    | Maybe                                    |
| Exploration –<br>Lunar, Mars,<br>Jupiter | Phasing orbits  | During<br>phasing<br>orbits                                 | During<br>phasing<br>orbits                                 | Yes              | Yes         | Possibly          | Yes                          | Maybe               | No                 | Yes                    | Yes                                      |

https://radhome.gsfc.nasa.gov/radhome/papers/SSPVSE05\_LaBel.pdf

# Two Example Missions

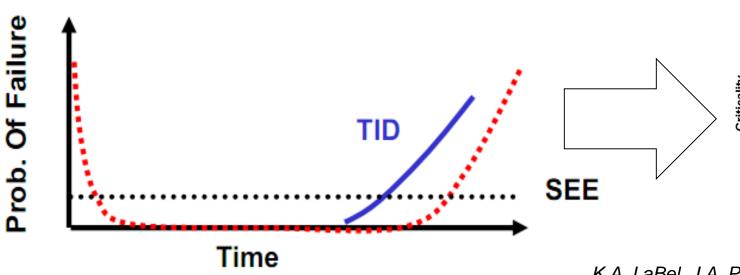


#### LEO Technology Demonstration

- SEE more of a driver than TID
- Un-vetted technology

### Interplanetary Asset

- Mission objectives
- Exotic environment at target



#### Environment/Lifetime

|          |        | Low   | Medium  | High  |
|----------|--------|---|---|---|
| (m)      | Low    | COTS upscreening/<br>testing optional; do<br>no harm (to others)  | COTS upscreening/<br>testing recommended;<br>fault-tolerance<br>suggested; do no<br>harm (to others)      | Rad hard<br>suggested. COTS<br>upscreening/<br>testing<br>recommended;<br>fault tolerance<br>recommended          |
| Cinconny | Medium | COTS upscreening/<br>testing<br>recommended; fault-<br>tolerance suggested  | COTS upscreening/<br>testing recommended;<br>fault-tolerance<br>recommended                               | Level 1 or 2, rad<br>hard suggested.<br>Full upscreening<br>for COTS. Fault<br>tolerant designs for<br>COTS.      |
|          | High   | Level 1 or 2<br>suggested. COTS<br>upscreening/<br>testing<br>recommended. Fault<br>tolerant designs for<br>COTS. | Level 1 or 2, rad hard<br>suggested. Full<br>upscreening for COTS.<br>Fault tolerant designs<br>for COTS. | Level 1 or 2, rad<br>hard<br>recommended. Full<br>upscreening for<br>COTS. Fault<br>tolerant designs for<br>COTS. |

K.A. LaBel, J.A. Pellish, "Notional Radiation Hardness Assurance (RHA) Planning For NASA Missions: Updated Guidance" HEART Conference 2014.

## RHA Risk Acceptance



- Define the Environment
  - External to the spacecraft
- Evaluate the Environment
  - Internal to the spacecraft
- Define the Requirements
  - Define criticality factors
- Evaluate Design/Components
  - Existing data/Testing
  - Performance characteristics
- "Engineer" with Designers
  - Parts replacement/Mitigation schemes
- Iterate Process
  - Review parts list based on updated knowledge

### LEO Tech Demo

# Interplanetary Asset



#### Environment/Lifetime

|             |        | Low   | Medium                                      | High   |  |
|-------------|--------|---|---|--|--|
|             | High   | Spenvis Run /<br>Dose Depth /                         | Dose depth evaluation at thinnest shielding | Ray-trace for subsystem                              |  |
| Criticality | Medium | Spenvis Run /<br>Dose Depth /                         | Spenvis Run /<br>Dose Depth /               | Dose depth<br>evaluation at<br>thinnest<br>shielding |  |
|             | Low    | Similar mission<br>dose, based<br>same solar<br>cycle | Spenvis Run /<br>Dose Depth                 | Spenvis Run /<br>Dose Depth /                        |  |

#### Environment/Lifetime

|             |        | Low  | Medium                        | High   |
|-------------|--------|--|-------------------------------|--|
|             | High   | Ray-trace for subsystem                              |                               |  |
| Criticality | Medium | Dose depth<br>evaluation at<br>thinnest<br>shielding | Ray-trace for subsystem       | Ray-trace for subsystem                              |
|             | Low    | Spenvis Run /<br>Dose Depth /                        | Spenvis Run /<br>Dose Depth / | Dose depth<br>evaluation at<br>thinnest<br>shielding |

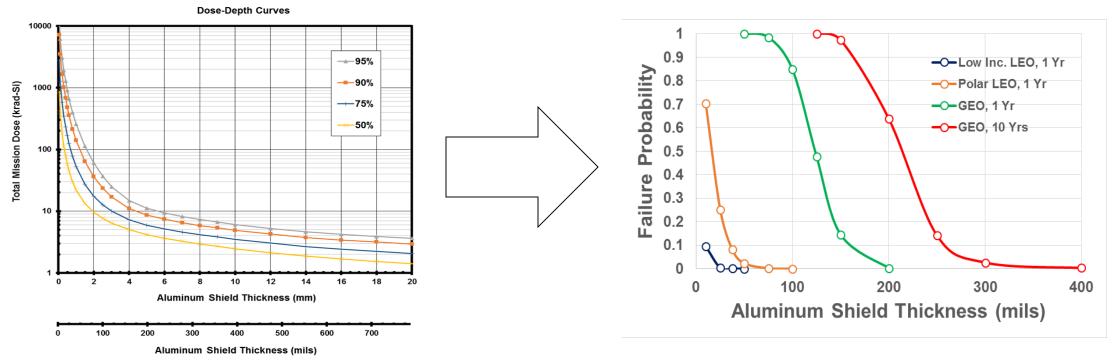
# New Considerations: NEPP Efforts to Improve RHA



- Define / Evaluate the Environment
  - Inclusion of Environment Variability
    - » M. Xapsos; C. Stauffer; A. Phan; S. McClure; R. Ladbury; J. Pellish; M. Campola; K. LaBel, "Inclusion of Radiation Environment Variability in Total Dose Hardness Assurance Methodology," in *IEEE Transactions on Nuclear Science*, vol.PP, no.99, pp.1-1.
- Define the Requirements
  - Requirements by Technology
    - » JESD57 updates, establishes testing procedures.
    - » NEPP RHA guideline & Small Mission RHA .
- Evaluate Design/Components and "Engineer" with Designers
  - Bayesian Methodologies
    - » R. Ladbury, J. L. Gorelick, M. A. Xapsos, T. O'Connor and S. Demosthenes, "A Bayesian Treatment of Risk for Radiation Hardness Assurance," 2005 8th European Conference on Radiation and Its Effects on Components and Systems, Cap d'Agde, 2005, pp. PB1-1-PB1-8.
    - » Ron Schrimpf's MRQW talk before the break.

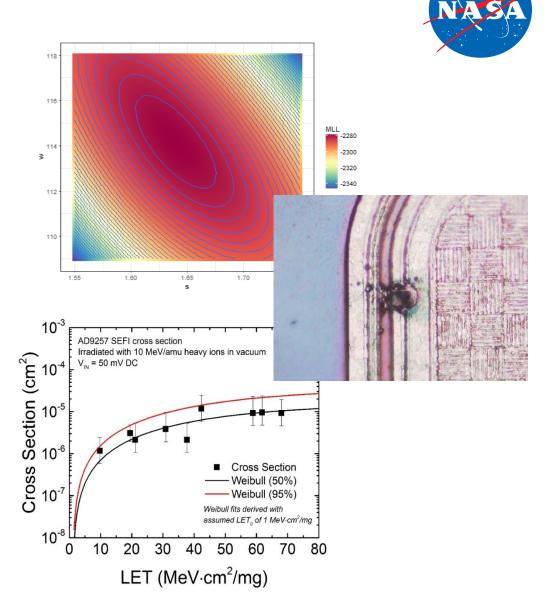
# Inclusion of Environment Variability

- NASA
- Confidence levels on environment external to the spacecraft account for variation.
- Transport to spacecraft's internal environment remains the same.
- Convolution of part failure distribution with environment confidence removes the ambiguity of RDM while maintaining/tailoring conservatism for TID/DD.



# Requirements by Technology

- SEL, SEB
  - Environment driven, risk avoidance
  - Diode Derating
- SEGR, SEDR
  - Effect driven, normally incident is worst case
  - SOA
  - Validate test procedures
- Proton SEE susceptible parts are evaluated as determined here: <a href="https://nepp.nasa.gov/files/25401/Proton\_RHAGuide\_NASAAug09.pdf">https://nepp.nasa.gov/files/25401/Proton\_RHAGuide\_NASAAug09.pdf</a>
- MBU, MCU, SEFI, Locked States only on devices that can exhibit the effect



# Summary



- Challenges identified in the past are here to stay
- RHA flow doesn't change, risk acceptance needs to be tailored
- Varied missions profiles and environments don't necessarily benefit from the same risk reduction efforts or cost reduction attempts
- We need data with statistical methods in mind
- Risks versus rewards can have big impact on mission enabling technologies

Sponsor: NASA Electronic Parts and Packaging (NEPP) Program



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### **THANK YOU**